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Receptivity of a
Cryogenic Coaxial
Liquid Jet to
Acoustic
Disturbances
ILASS-Americas 2014

Jeff Wegener, UCLA
David Forliti, Sierra Lobo, Inc.
Ivett Leyva, AFRL/RQRE
Doug Talley, AFRL/RQRC





Outline



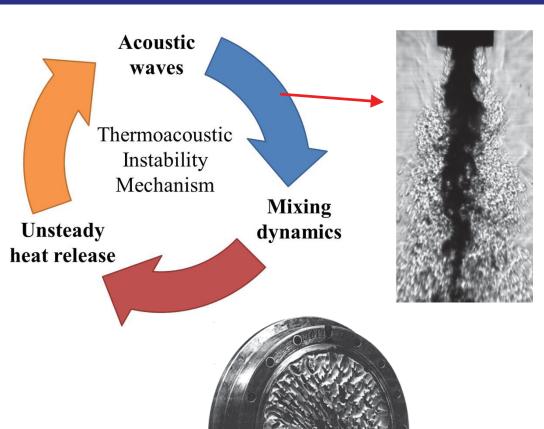
- Motivation and objectives
- Parameters of the forced coaxial jet
- Experimental facility
- Results
 - Unforced cases
 - Pressure node/antinode forcing
- Conclusions





Motivation: Combustion Instability in Rocket Engines











Objectives



Investigate acoustic receptivity characteristics of a model liquid rocket engine injector

- Dimensionless frequency
- Acoustic amplitude
- Momentum flux ratio
- Location within the mode

"Preferred mode" of the coaxial jet

- Definition of natural frequency of the flow
- Characteristic velocity scale



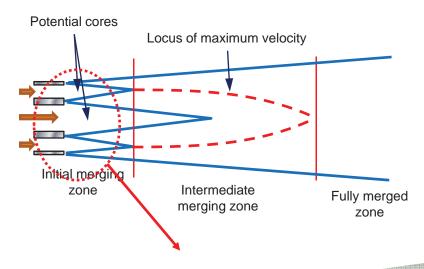


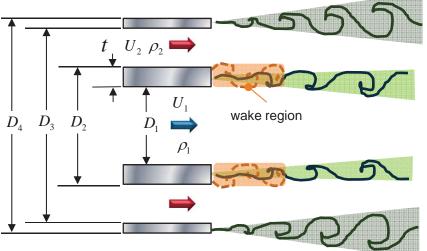
The Coaxial Jet

Outer shear layer

Inner shear layer







Geometry parameters

Area ratio

Dimensionless post thickness

$$AR = \frac{D_3^2 - D_2^2}{D_1^2}$$

$$\frac{t}{D_1}$$

Flow parameters

$$Re_i = \frac{\rho_1 U_1 D_1}{\mu_1} \qquad Re_i =$$

$$Re_{i} = \frac{\rho_{2}U_{2}(D_{3} - D_{2})}{\mu_{2}}$$

$$Re_{i} = \frac{\rho_{1}U_{1}D_{1}}{\mu_{1}} \qquad Re_{i} = \frac{\rho_{2}U_{2}(D_{3} - D_{2})}{\mu_{2}}$$
$$J = \frac{\rho_{2}U_{2}^{2}}{\rho_{1}U_{1}^{2}} \quad r = \frac{U_{2}}{U_{1}} \qquad s_{1} = \frac{\rho_{2}}{\rho_{1}} \quad s_{2} = \frac{\rho_{3}}{\rho_{2}}$$

$$We = \frac{\rho_2 U_2^2 D_1}{\sigma}$$

Inflow boundary conditions

- Mean velocity profiles
- RMS fluctuation profiles
- Spectral content



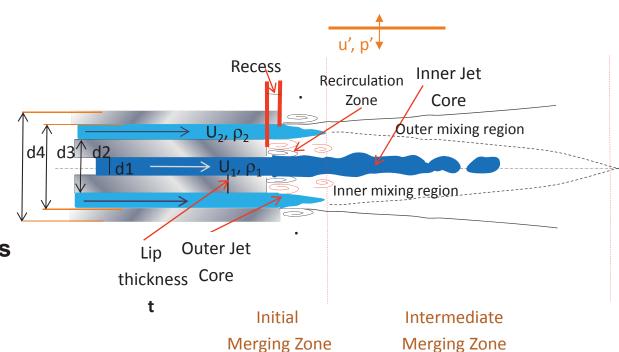


The Forced Coaxial Jet



Very low density ratio regime: $0.005 < \frac{\rho_2}{\rho_1} < 0.1$

- Transverse Acoustic mode from chamber/siren
 - f=f(c, geometry)
- 2. Acoustic modes propellant lines
 - f~c/2L
- 3. Post wake
 - St=ft/U_{ch}
- 4. Shear layer instabilities
 - $St_{\theta} = f\theta/U_{ch}$
- 5. Jet preferred modes
 - $St=fD_{ij}/U_{ij}$







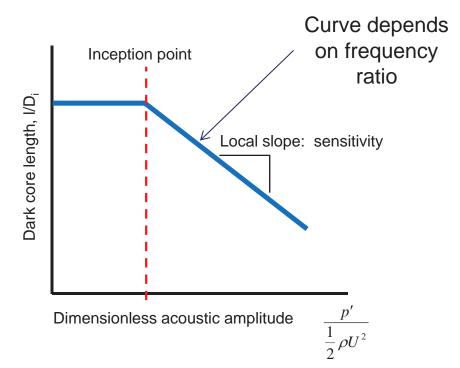
Forcing Characterization

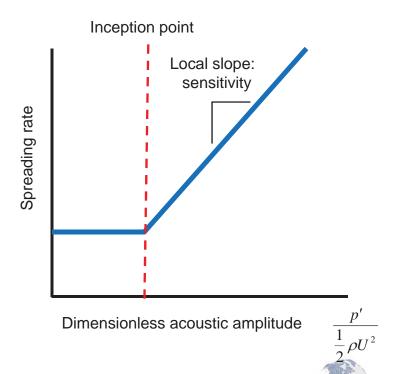


- Shift pressure normalization from chamber pressure to injector dynamic pressure
- Normalize the frequency by the preferred mode of the coaxial jet
- Identify receptivity inception point—threshold for coupling between acoustics and flame

$$P'/\overline{P}_c \rightarrow \frac{P'}{\rho U^2/2}$$

$$F = \frac{f_{forcing}}{f_{jet}}$$

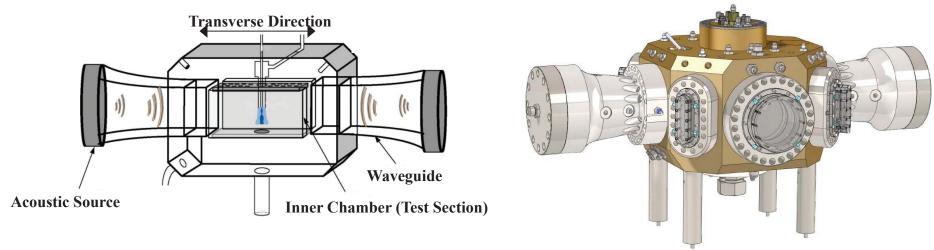






Experimental Facility





Capabilities

- Cryogenic propellant temperature control with high accuracy (±1 K)
- Sub- and super-critical chamber pressure (p_c up to 10.4 MPa)
- High amplitude acoustic forcing ($p'/p_c \sim 0.02$)
- Coaxial injector with extended length for fully developed turbulent flow ($I_e/D > 110$)
- High-speed diagnostic tools
 - Pressure transducer(s) natural frequency > 100 kHz
 - Time-series backlit imaging (f > 25 kHz)
 - Off-axis windows for future PIV/PLIF measurements



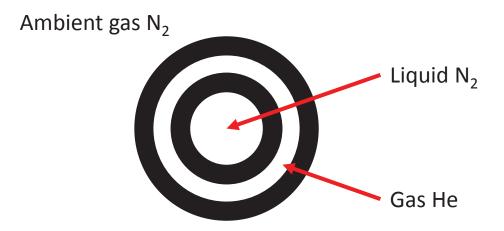


Experimental Conditions



New injector

- $D_1 = 1.4 \text{ mm}$
- -AR = 1.68
- $t/D_1 = 0.27$
- •J = 2 and 6
- •N₂ inner jet @ 120 K
- •Gaseous He @ 275 K
- $\cdot Re_1 \sim 1.5 \times 10^4$
- • $Re_2 \sim 1 \times 10^4$
- Fully-developed turbulent flow conditions
- •Chamber pressure 2.8 MPa (400 psi)→ subcritical

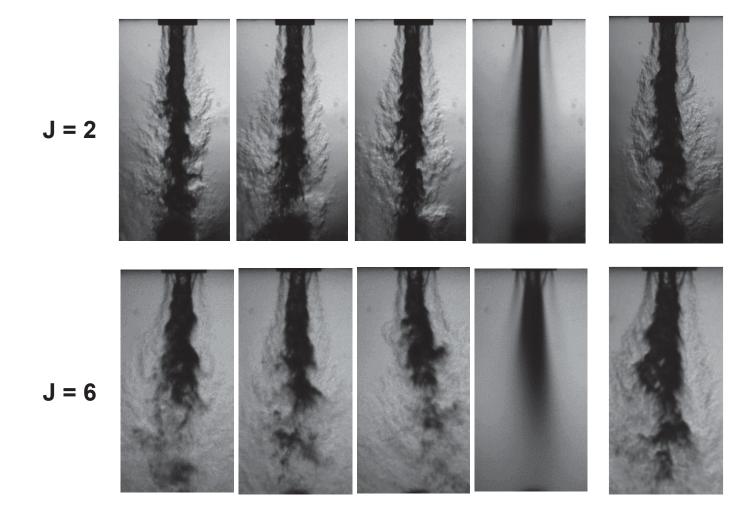






Unforced Cases



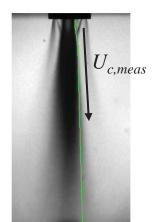






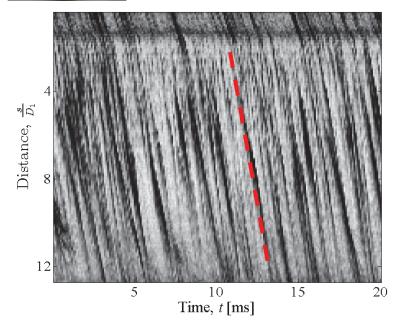
Convection Velocity

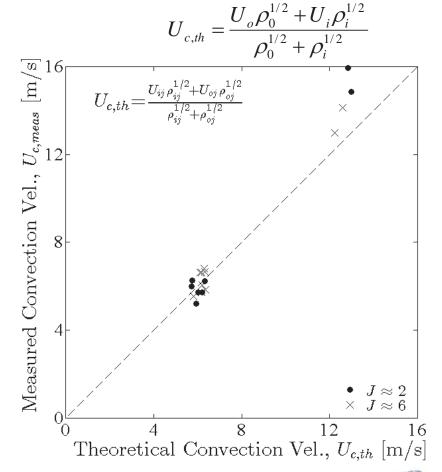




Verify the accuracy of the Dimotakis (1986) expression for shear layer convection velocity for these flow conditions.

$$U_{c,meas} = \frac{\Delta s}{\Delta t}$$

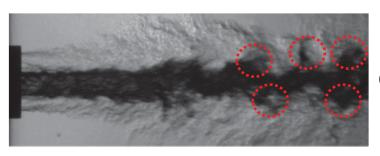






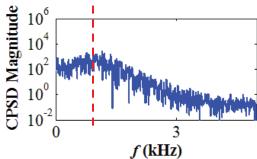
Preferred Mode Frequency

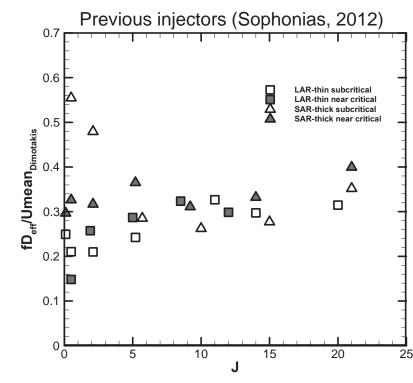


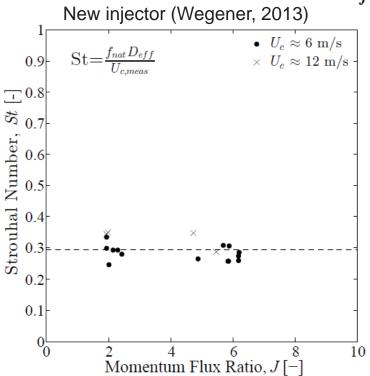


Most energetic convective mode pair from POD











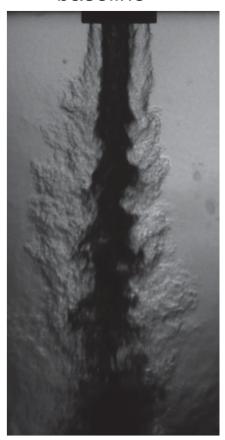


Forced Cases

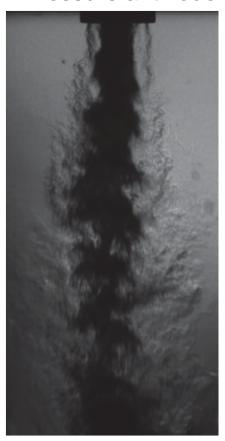


Representative cases for pressure node and pressure antinode

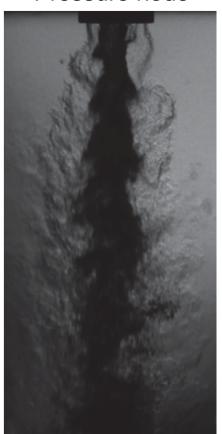
baseline



Pressure antinode



Pressure node

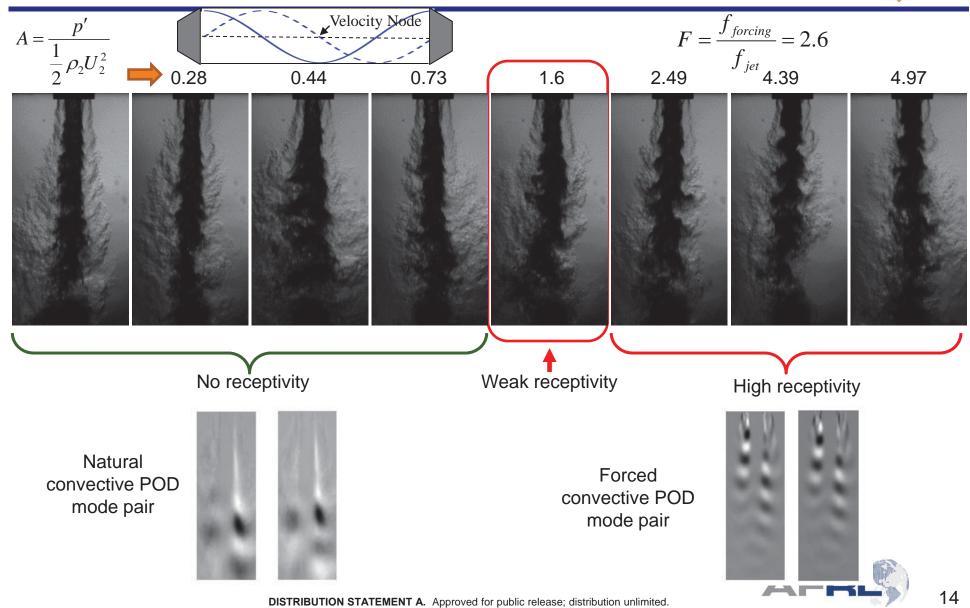






Pressure Antinode Response J = 2

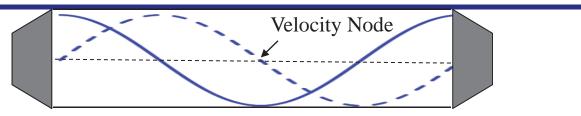






Pressure Antinode Mechanism







Outer jet mass flow pulsations

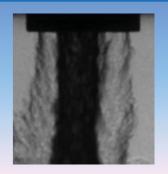
POD structure

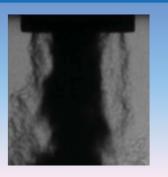
unforced

Max forcing

POD structure

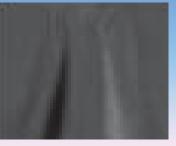




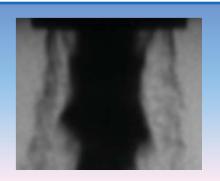










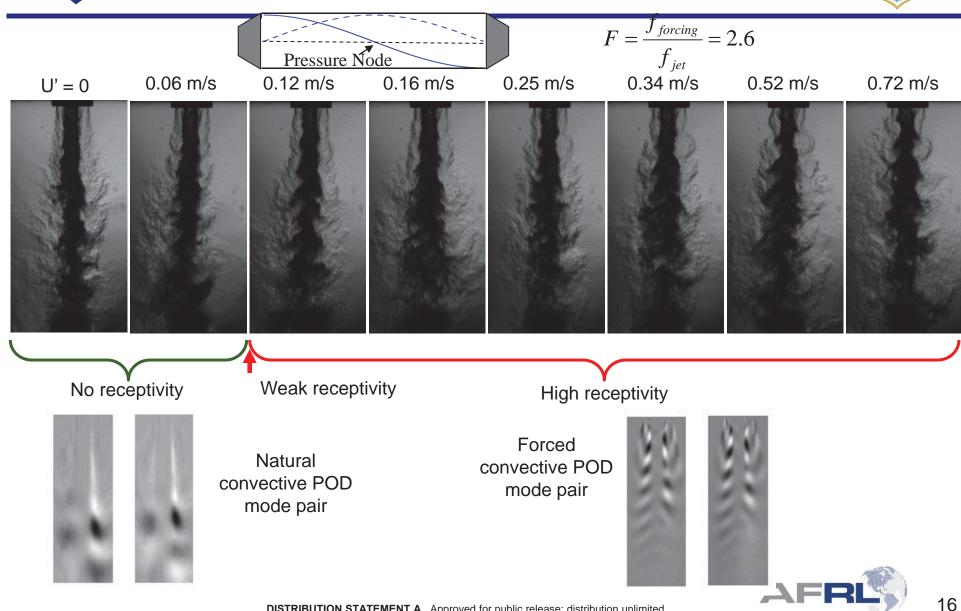






Pressure Node Response J = 2

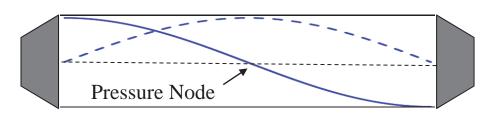




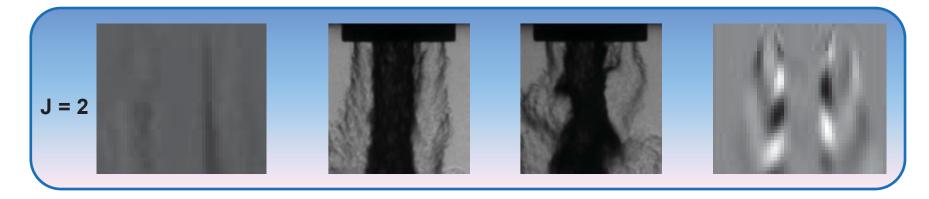


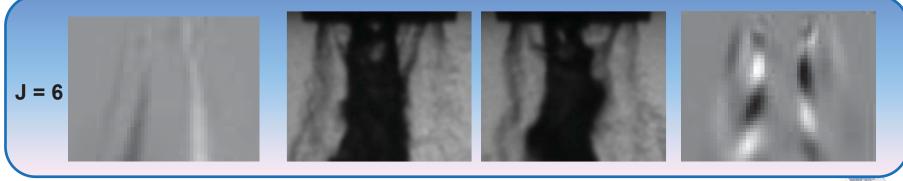
Pressure Node Mechanism





Apparent excitation of antisymmetric mode in the outer jet that drives instabilities in the inner jet

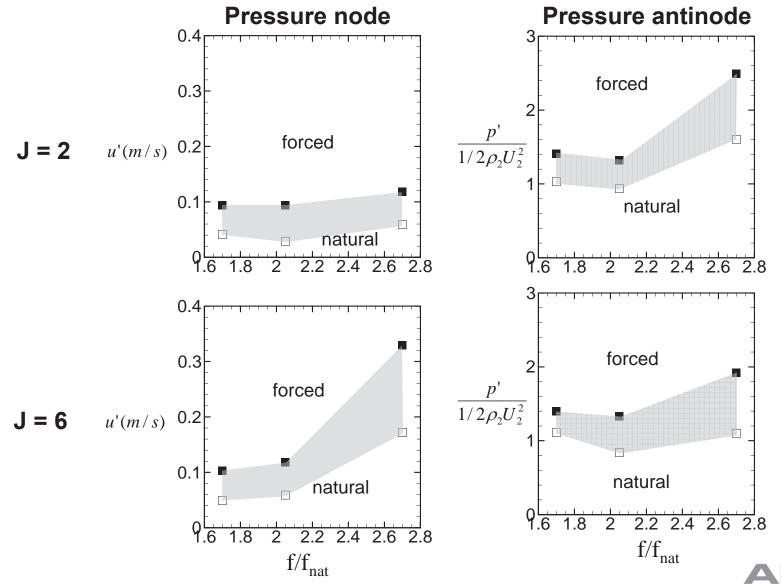






Receptivity







Summary



- Convection velocity predicted using shear layer model
- Coaxial jet preferred mode scaling law
- Receptivity characteristics for J = 2 and 6
 - Pressure antinode → outer jet puffing mechanism
 - Scales with outer jet dynamic pressure
 - Pressure node → excitation of helical or antisymmetric mode
 - Very sensitive mode—driven by low level forcing





Future Work



- Determine robustness of scaling laws
 - Convection velocity (i.e. Dimotakis law)
 - Strouhal number
- Supercritical conditions
- Reacting flow conditions
- Multiple injectors





Backup slides



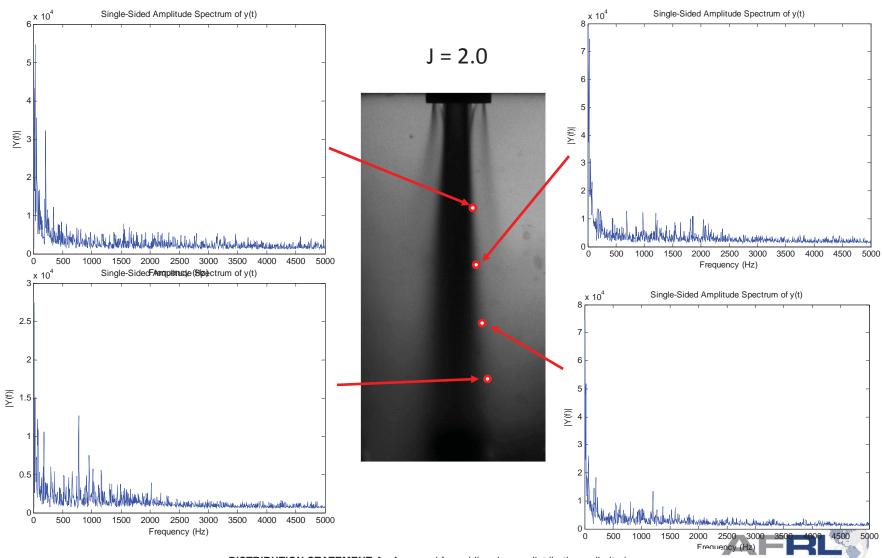




Unforced Coaxial Jets



•Frequency depends on location

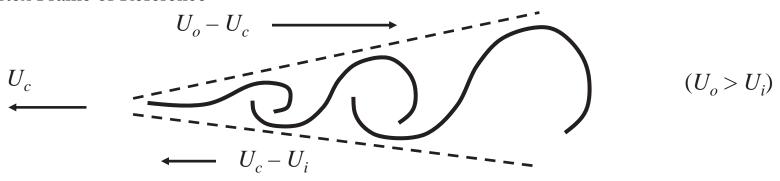




Convection Velocity



Convective Shear Layer Velocity by Dimotakis (1986) Vortex Frame of Reference



- Bernoulli's equation
 - A stagnation point must exist between vortices. Therefore, along a line through this point, dynamic pressures are approximately equal.

$$\begin{aligned}
\rho_o(U_o - U_c)^2 &\approx \rho_i (U_i - U_c)^2 \\
U_c &= \frac{U_o \rho_o^{1/2} + U_i \rho_i^{1/2}}{\rho_o^{1/2} + \rho_i^{1/2}}
\end{aligned}
St = \frac{f_{nat} D}{U_c}$$



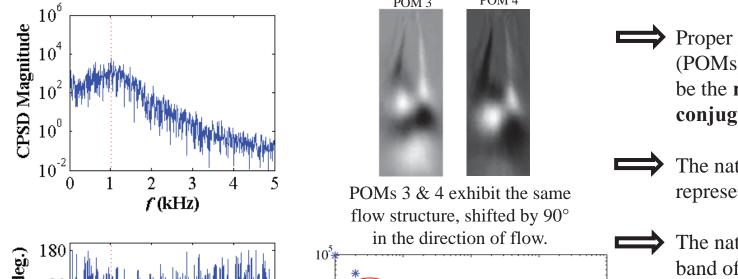
If St, D, U_c are held constant then f_{nat} may be constant.

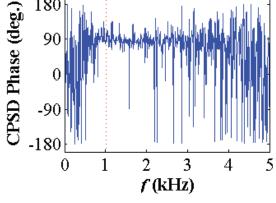


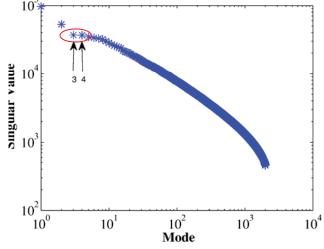
Convective Mode from POD



- Proper Orthogonal Decomposition
 - To identify traveling, coherent structures, a conjugate mode pair is identified as any two modes whose CPSD magnitude peaks near a phase of $\pm 90^{\circ}$.¹²







Proper orthogonal modes
(POMs) 3 & 4 were found to
be the most energetic
conjugate pair.

The natural mode is represented by POMs 3 & 4.

The natural mode spans a band of frequencies rather than a single peak frequency.

Arienti, M. and Soteriou, M.C.(2009)

